

# SSC-JE

# 2025

**Staff Selection Commission**  
Junior Engineer Examination

## **Mechanical Engineering**

### **Strength of Materials**

Well Illustrated **Theory** *with*  
**Solved Examples** and **Practice Questions**



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# Strength of Materials

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# Properties of Metals, Simple Stress-Strain and Elastic Constants

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## 1.1 Introduction

- Strength of material is a branch of applied mechanics that deals with the behaviour of solid bodies subjected to various types of loading and internal forces developed due to these loading. A thorough understanding of mechanical behaviour is essential for the safe design of all structures, whether buildings, bridges, machines motors, submarines or airplanes. Hence, strength of material is a basic subject in many engineering fields.
- So the objective of our analysis will be to determine the stresses, strains and deflections produced by the loads in different structures. Theoretical analysis and experimental results have equally important role in the study of strength of materials so these quantities are found for all values of load upto the failure load, then we will have a complete picture of the mechanical behaviour of the body.

## 1.2 Normal Stress

- It is the internal resistance offered by the body against external loading or deformation which is force per unit area.
- Stress induced in the material depends upon the nature of force, point of application and cross-section area of material. Stress can be tensile or compressive in nature depending on the nature of load.

$$\sigma = \frac{P}{A} \text{ N/mm}^2 \text{ or MPa}$$

- Sign convention:           Tensile stress = +ve  
  Compressive stresses = -ve
- Stresses are induced only when motion of bar is restricted either by some force or reaction induced. If body or bar is free to move or free expansions allowed then no stresses will be induced.
- Pressure has same unit but pressure is different physical quantity than stress. Pressure is external normal force distributed over surface.
- On the basis of cross-section area considered during calculation of stresses. Direct stresses can be of following two types.
  1. Engineering stresses or nominal stresses.
  2. True stresses or actual stresses

**Mathematically:**

Engg. stress,  $\sigma = \frac{P}{A_0}$

$A_0$  = Original cross-section area of specimen taken

True stress,  $\sigma_t = \frac{P}{A_a}$

$A_a$  = Actual cross-section area of specimen at any time of loading i.e., changed area of cross-section due to loading.

$$A_a = A_0 + \Delta A$$

- In tension,  $A_a < A_0$  ( $\because L_a > L_0$  (as  $L_a = L_0 + \Delta L$ ) and  $A_a L_a = A_0 L_0$ )
- $\therefore \sigma_t > \sigma$
- In compression,  $A_a > A_0$
- $\therefore \sigma_t < \sigma$

**1.3 Strain**

- An axially loaded bar undergoes a change in length, becoming longer when in tension and shorter when in compression. Thus, the elongation or shortening in axially loaded member per unit length is known as strain.

**Mathematically:**

$$\epsilon = \frac{\delta L}{L}$$

- Strain is dimensionless quantity.
  - If member is in tension, strain is called tensile strain.
  - If member is in compression, then strain is called compressive strain.
  - On the basis of length of member used in calculation of strain, strain can be of following two types.
    - Engineering or nominal strain
    - True or actual strain
1. **Engineering or nominal strain**

$$\epsilon = \frac{\delta L}{L_0}$$

$L_0$  = Original length of member

**2. True or actual strain**

$$\epsilon_T = \frac{\Delta L}{L_a}$$

$L_a$  = Actual length of member

$$L_a = L_0 + \Delta L$$

$\oplus$  = Tension,  $\ominus$  = compression.

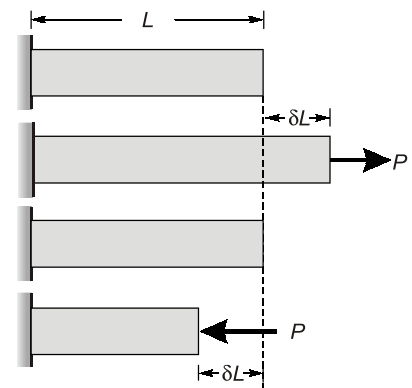


Figure 1.1

## 1.4 Stress-strain Curve

### 1.4.1 Simple Tension Test for Mild Steel

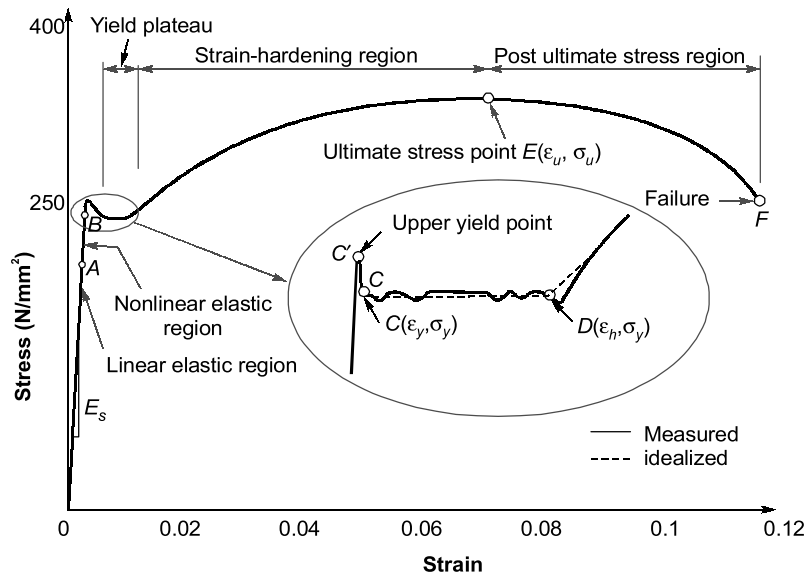


Figure 1.2

- **A is Limit of proportionality:** Beyond A linear variation ceases. Hook's law is valid in OA.
- **B is Elastic limit:** The maximum stress up to which a specimen regains its original length on removal of applied load. For mild steel B is very near to A. However, for other materials B may be greater than A.
- **C' is Upper yield point:** The magnitude of the stress corresponding to C' depends on the cross-sectional area, shape of the specimen and the type of the equipment used to perform the test. It has no practical significance.
- **C is Lower yield point:** The stress at C is the yield stress ( $\sigma_y$ ) with a typical value of  $\sigma_y = 250 \text{ N/mm}^2$  for mild steel. The yielding begins at this stress.
- **CD represents perfectly plastic region:** It is the strain which occurs after the yield point C, without any increase in stress. The strain corresponding to point D is about 1.4% and corresponding to C is about 0.12% for mild steel. Hence, plastic strain is 10 to 15 times of elastic strain.
- **DE represents strain hardening:** In this range further addition of stress gives additional strain. However, strain increases with faster rate in this region. The material in this range undergoes change in its atomic and crystalline structure, resulting in increased resistance to further deformation. This portion is not used for structural design.
- **E is Ultimate point:** The stress corresponding to this point is ultimate stress ( $\sigma_u$ ) and the corresponding strain is about 20% for mild steel.
- **F is fracture point:** Stress corresponding to this is called breaking stress and strain is called fracture strain. It is about 25% for mild steel.
- Region between E and F is the necking region, in which area of cross-section is drastically decreased.

**Do you know?** Strain that occurs before the yield point is called elastic strain and that which occurs after yield point with no increase in stress is called plastic strain. For mild steel, plastic strain is 10 to 15 times of elastic strain.

**Example 1.1**

A rod of dimension 20 mm × 20 mm is carrying an axial tensile load of 10 kN.

The tensile stress developed is

- (a) 0.025 MPa      (b) 0.25 MPa      (c) 25 MPa      (d) 250 MPa

[SSC JE : 2018]

**Solution: (c)**

$$\sigma = \frac{P}{A} = \frac{10 \times 1000}{20 \times 20} = 25 \text{ MPa}$$

**Example 1.2**

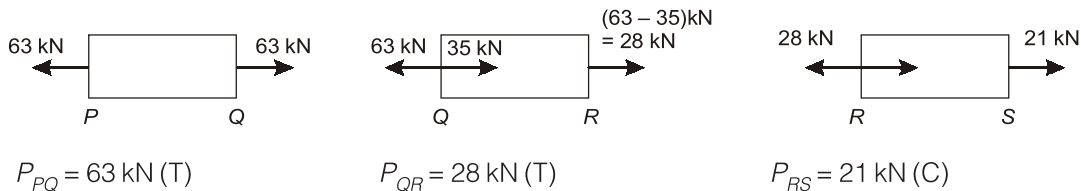
A cross-section bar of area 700 mm<sup>2</sup> is subjected to an axial load as shown in the figure below, what is the value of stress (MPa) in the section QR?



- (a) 30      (b) 40      (c) 50      (d) 60

[SSC JE : 2018]

**Solution: (b)**



$$\sigma_{QR} = \frac{P_{QR}}{A} = \frac{28000}{700} = 40 \text{ MPa}$$

- Ideal curve for tension is shown in the figure. However, actual behaviour is different and indicates apparently reduced yield stress in compression for mild steel. The divergence between tension and compression results is explained by Baughinger and is called **Baughinger effect**.
- The stress defined as the ratio of load to original area ( $A_0$ ) is known as 'engineering stress' or 'stress' or 'nominal stress'.

$$\text{Engineering stress or Nominal stress} = \frac{P}{A_0}$$

- On the contrary when changing area is taken into account, the ratio of load to actual area ( $A$ ) is called 'true stress'.

$$\text{True stress} = \frac{P}{A}$$

### 1.4.2 Actual Curve Vs. Engg. Curve in Tension

**NOTE**


- The fracture strain depends on % carbon present in steel.
- With increase in % carbon, fracture strain reduces.
- With increase in carbon content, steel has higher yield stress and higher ultimate stress.

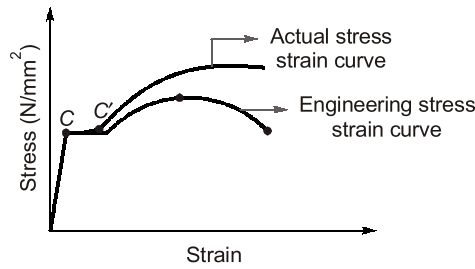


Figure 1.3

### 1.4.3 Compression Curve for Mild Steel

- In compression, engineering stress-strain curve lies above the actual stress-strain curve as shown in figure.
- In compression mild steel has yield stress  $\sigma_y = 263 \text{ N/mm}^2$ , slightly greater than tension.
- **Stress-strain curve for other grades of steel in tension.**
- Among all steel grades, high tension steel (HTS) is more brittle and mild steel is more ductile.
- High tension steel has higher ultimate strength than other grade of steel.

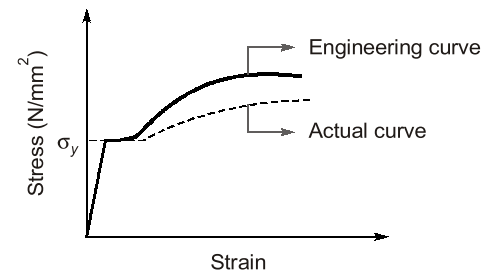


Figure 1.4

## 1.5 Properties of Metals

### 1.5.1 Ductility

- Ductility is the property by which material can be stretched. Large deformations are thus possible in ductile materials before the absolute failure or rupture takes place. Some of the examples are mild steel, aluminium, copper, manganese, lead, nickel, brass, bronze, monel metal etc.

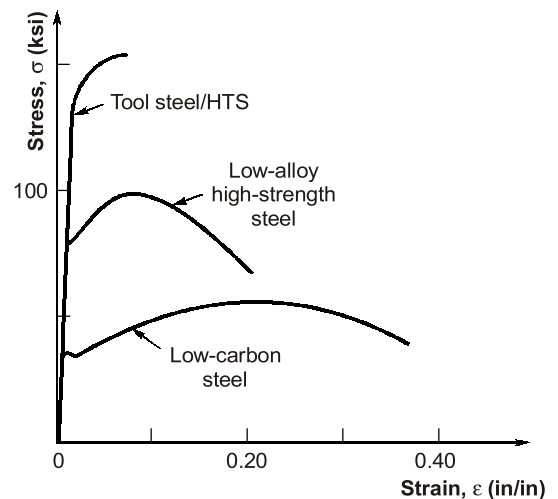


Figure 1.5

**Example 1.3**

The percentage reduction in area in case of cast iron when it is subjected to tensile test is of the order of

- (a) 0%                      (b) 10%                      (c) 20%                      (d) 25%

[SSC JE: 2018]

**Solution: (a)**

The % reduction in area in case of cast iron when it is subjected to tensile test is of the order of 0%, because cast iron being brittle material does not undergo elongation.

### 1.5.2 Brittleness

- Brittleness is the lack of ductility i.e. material cannot be stretched. In brittle materials, failure takes place with a relatively smaller deformation. This property is undesirable. For brittle materials fracture

point and ultimate points are same, and after proportional limit very small strain is seen. Some of the examples are cast iron, concrete and glass.

- To distinguish between these two types of materials, materials with strain less than 5% at fracture point are regarded as brittle and those having strains greater than 5% at fracture point are called ductile (this value for mild steel at fracture is about 25%).

### 1.5.3 Malleability

- The property by which a material can be uniformly extended in a direction without rupture. A malleable material possess a high degree of plasticity. This property is of great use in operations like forging, hot rolling, stamping etc.

### 1.5.4 Toughness

- The property which enables material to absorb energy without fracture. This property is very desirable in case of cyclic loading or shock loading.
- The **modulus of toughness** is measured as area under entire stress-strain curve and is the energy absorbed by material of the specimen per unit volume upto fracture stage.

$$\text{Modulus of toughness} = \left[ \frac{\sigma_y + \sigma_u}{2} \right] \epsilon_f$$

Where,

$\sigma_y$  = Yield tensile strength

$\sigma_u$  = Ultimate tensile strength

$\epsilon_f$  = Strain at fracture point

- The modulus of toughness will depend upon ultimate tensile strength and strain at failure (fracture strain). Hence the material which is very ductile will exhibit a higher modulus of toughness as the case with mild steel.
- The **modulus of resilience** is the maximum elastic energy per unit volume that can be absorbed without attaining plastic stage.

$$\text{Modulus of resilience } (u) = \frac{\sigma_y^2}{2E}$$

- **The modulus of resilience** depends upon yield strength and hence a material with higher yield strength will have higher modulus of resilience.
- Higher toughness is a desirable property in materials used for gears, chains, crane hooks, freight car etc. Higher resilience is desirable in springs.

### 1.5.5 Hardness

- Hardness is defined as the resistance to indentation or scratching or surface abrasion.
- There are two methods of hardness measurement:
  1. Scratch hardness - commonly measured by Mohr's test.
  2. Indentation hardness (abrasion) measured by
    - Brinell hardness method
    - Rockwell hardness
    - Vickers hardness
    - Knoop hardness

*It should be noted that ductile materials are tough and brittle materials are hard.*



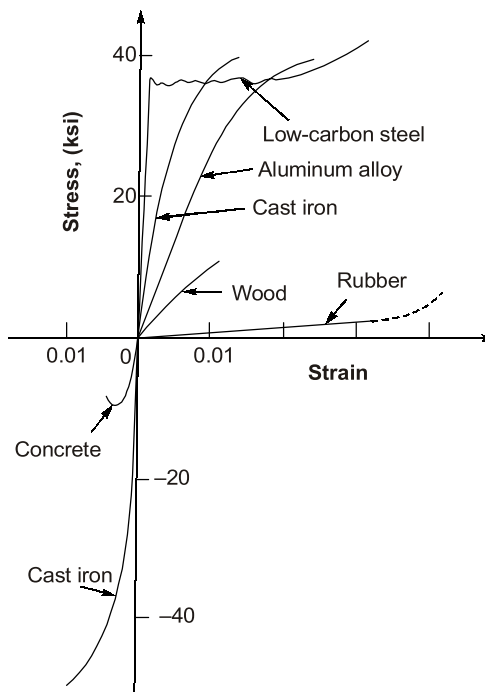
### 1.5.6 Fatigue

- It has been found that material behave differently under the static loading and dynamic loading.
- The behaviour of material under variable loads (dynamic loads) is referred to as **fatigue**. In recent past several failures of structures have been noted due to fatigue.
- Factors affecting fatigue are:
  1. Loading conditions
  2. Frequency of loading
  3. Corrosion
  4. Temperature
  5. Stress concentration

### 1.5.7 Creep and Stress Relaxation

- At any temperature, a material will progressively deform with the passage of time under constant loading, even if the stress is below yield point. This phenomenon is called creep. However, such deformation is negligibly small at lower temperature.
- At higher temperature, due to greater mobility of atoms, most of the materials loose their strength and elastic constants also get reduced. Hence, greater deformations at elevated temperature results even under constant loading. Therefore, creep is more pronounced at higher temperature, and thus it must be considered for design of engines and furnaces.
- The temperature at which the creep becomes very appreciable is half of the melting point temperature on absolute scale and is known as **homologous temperature**.
- If a wire of metal is stretched between two immovable supports, so that it has an initial tension stress  $\sigma_0$ . The stress in the wire gradually diminishes, eventually reaching a constant value. This process, which is a manifestation of creep, is called **stress relaxation**.

**Do you know?** For metallic minerals creep becomes an important consideration at half of the melting point temperature on absolute scale.



**Figure 1.6 :** Stress-strain diagram for different material

## 1.6 Hooke's Law

- It states that upto the proportional limit stress is proportional to strain. i.e.

$$\text{Stress} \propto \text{strain}$$

$$\sigma = E\epsilon$$

$$\frac{\text{Stress}}{\text{Strain}} = \text{Constant} = E$$

Where,  $E$  = Young's modulus of elasticity

- Hooke's law is valid upto limit of proportionality. However for mild steel proportional limit and elastic limit are almost equal. But for other metals and materials elastic limit may be higher than proportional limit. e.g. rubber.
- The slope of stress-strain curve is called modulus of elasticity ( $E$ ). The modulus of elasticity ( $E$ ) is the constant of proportionality which is defined as the intensity of stress that causes unit strain. Thus, modulus of elasticity ( $E$ ) has the units same as units of stress.

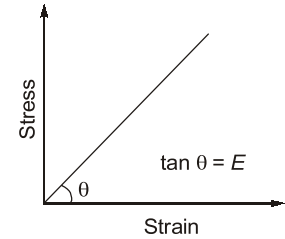


Figure 1.7

### Assumptions in Hooke's Law:

1. Material is homogeneous (properties are equal at all points)
2. Material is isotropic (properties are equal in all direction)
3. Material is elastic

#### NOTE



1. If properties are different in three mutually perpendicular direction then material is called Orthotropic material.
2. If properties are different in all direction then material is called Non-isotropic or an anisotropic, e.g., Crystal.

## 1.7 Proof Stress

- When a material such as aluminium does not have an obvious yield point and yet undergoes large strains after the proportional limit is exceeded, an arbitrary yield stress may be determined by the offset method.
- A line parallel to initial linear part is drawn, which is offset by some standard amount of strain such as 0.2%. The intersection of the off set point ( $A$ ) defines the yield stress or off set yield stress, which is slightly above the proportional limit and is called **proof stress**.

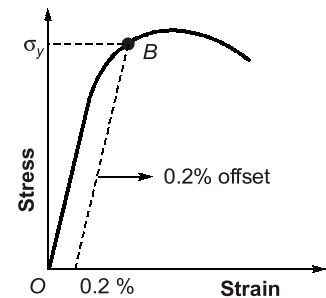


Figure 1.8

## 1.8 Saint Venant Principle

- It states that the equation  $\sigma = P/A$  gives the axial stress on a cross-section only when the cross-section is atleast a distance ' $b$ ' away from any concentrated load or dia-continuity in shape, where ' $b$ ' is the largest lateral dimension of the bar.
- Let  $\sigma_{\text{avg}}$  is the average direct stress at any section, then

$$\begin{aligned} &\Rightarrow \sigma_{\text{max}} \text{ at } 1-1 = 1.387 \sigma_{\text{avg}} \\ &\Rightarrow \sigma_{\text{max}} \text{ at } 2-2 = 1.027 \sigma_{\text{avg}} \\ &\Rightarrow \sigma_{\text{max}} \text{ at } 3-3 = \sigma_{\text{avg}} \end{aligned}$$